

Text Book on Dynamic Relaxation Method



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Dynamic Relaxation Method
Theoretical Analysis, Solved Examples
And Computer Programming

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Dedication

In the name of Allah, the merciful, the compassionate

All praise is due to Allah and blessings and peace is upon his messenger and servant, **Mohammed**, and upon his family and companions and whoever follows his guidance until the day of resurrection.

To the memory of my mother **Khadra Dirar Taha**, my father **Mohammed Elmardi Suleiman**, and my dear aunt **Zaafaran Dirar Taha** who they taught me the greatest value of hard work and encouraged me in all my endeavours.

To my first wife **Nawal Abbas** and my beautiful three daughters **Roa, Rawan** and **Aya** whose love, patience and silence are my shelter whenever it gets hard.

To my second wife **Limya Abdullah** whose love and supplication to Allah were and will always be the momentum that boosts me through the thorny road of research.

To Professor. **Mahmoud Yassin Osman** for reviewing and modifying the manuscript before printing process.

This book is dedicated mainly to undergraduate and postgraduate students, especially mechanical and civil engineering students plus mathematicians and mathematics students where most of the applications are of mathematical nature.

To Mr. **Osama Mahmoud** of Daniya Center for printing services whose patience in editing and re – editing the manuscript of this book was the momentum that pushed me in completing successfully the present book.

To my friend professor **Elhassan Mohammed Elhassan Ishag**, Faculty of Medicine, University of Gezira, Medani, Sudan.

To my friend **Mohammed Ahmed Sambo**, Faculty of Engineering and Technology, Nile Valley University, Atbara, Sudan.

To my homeland, Sudan, hoping to contribute in its development and superiority.

Finally, may Allah accepts this humble work and i hope that it will be beneficial to its readers

Acknowledgement

I am grateful and deeply indebted to Professor **Mahmoud Yassin Osman** for valuable opinions, consultation and constructive criticism, for without which this work would not have been accomplished.

I am also indebted to published texts in the analysis of dynamic relaxation method which have been contributed to the author's thinking. Members of Mechanical Engineering Department at Faculty of Engineering and Technology, Nile Valley University, Atbara – Sudan, and Sudan University of Science & Technology, Khartoum – Sudan have served to sharpen and refine the treatment of my topics. The author is extremely grateful to them for constructive criticisms and valuable proposals.

I express my profound gratitude to Mr. **Osama Mahmoud** of Daniya Center for computer and printing services, Atbara, who spent several hours in editing, re – editing and correcting the present manuscript.

Special appreciation is due to the British Council's Library for its quick response in ordering the requested bibliography, books, reviews and papers.

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Symbols and Abbreviations

$KS = K^*$

$NMAX$ = maximum number of iterations

EQ = equal to

Dx = discretization of solution in the x – direction

GT = greater than

LT = less than

$\frac{\partial u}{\partial t}$ = velocity

$\frac{\partial^2 u}{\partial t^2}$ = acceleration

Δt = time increment

ρ = inertia effect

k = damping effect

Preface

The method of dynamic relaxation in its early stages of development was perceived as a numerical finite difference technique. It was first used to analyze structures, then skeletal and cable structures, and plates. The method relies on a discretized continuum in which the mass of the structure is assumed to be concentrated at given points (i.e. nodes) on the surface. The system of concentrated masses oscillates about the equilibrium position under the influence of out of balance forces. With time, it comes to rest under the influence of damping. The iterative scheme reflects a process, in which static equilibrium of the system is achieved by simulating a pseudo dynamic process in time. In its original form, the method makes use of inertia term, damping term and time increment.

Chapter one includes introduction to dynamic relaxation method (DR) which is combined with finite differences method (FD) for the sake of solving ordinary and partial differential equations, as a single equation or as a group of differential equations. In this chapter the dynamic relaxation equations are transformed to artificial dynamic space by adding damping and inertia effects. These are then expressed in finite difference form and the solution is obtained through iterations.

In chapter two the procedural steps in solving differential equations using DR method were applied to the system of differential equations (i.e. ordinary and/ or partial differential equations). The DR program performs the following operations: Reads data file; computes fictitious densities; computes velocities and displacements; checks stability of numerical computations; checks convergence of solution; and checks wrong convergence. At the end of this chapter the dynamic relaxation (DR) numerical method coupled with the finite differences discretization technique is used to solve nonlinear ordinary and partial differential equations. Subsequently, a FORTRAN program is developed to generate the numerical results as analytical and/ or exact solutions.

Chapter three discusses the importance of using numerical methods in solving differential equations and stresses the use of dynamic relaxation technique in these applications.

The book is suitable as a textbook for a first course on dynamic relaxation technique in civil and mechanical engineering curricula. It can be used as a reference by engineers and scientists working in industry and academic institutions.

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CHAPTER ONE

Introduction

1.1 General Introduction:

Differential equations can describe nearly all systems undergoing change. They are ubiquitous in science and engineering as well as economics, social science, biology, business, health care, etc. Many mathematicians have studied the nature of these equations for hundreds of years and there are many well developed solution techniques. Often, systems described by differential equations are so complex, or the systems that they describe are so large, that a purely analytical solution to the equations is not tractable. It is in these complex systems where computer simulations and numerical methods are useful.

The techniques for solving differential equations based on numerical approximations were developed before programmable computers existed. During world war two, it was common to find rooms of people (usually women) working on mechanical calculators to numerically solve systems of differential equations for military calculations. Before programmable computers, it was also common to exploit analogies to electrical systems to design analogue computers to study mechanical, thermal, or chemical systems. As programmable computers have increased in speed and decreased in cost, increasingly complex systems of differential equations can be solved with simple programs written to run on a common PC. Currently, the computer on your desk and laptop computers can tackle problems that were inaccessible to the fastest supercomputers just 45 or 50 years ago. Therefore, the history of numerical solution of ordinary and partial differential equations is much younger than that of analytical solution methods, but the development of high speed computers nowadays makes the advent of numerical methods very fast and productive. On the other hand, the numerical approximation of ordinary and partial differential equations often demands a knowledge of several aspects of the problem,

such as the physical background of the problem in order to understand and interpret the behavior of expected solutions, or the algorithmic aspects concerned with the choice of the numerical method and the accuracy that can be achieved.

The aim of this text book is to discuss some modeling problems and provide the students with the knowledge of Dynamic Relaxation (DR) techniques for the numerical approximation of the model equations. The theory and application of dynamic relaxation method is a very nice combination of mathematical theory with aspect of implementation, modeling, and applications. So – called adaptive methods enable on one hand the prescription of a tolerance for the approximation, while on the other hand they make computations possible in cases where, for example, a uniformly refined mesh would be prohibitively costly even on nowadays' computers, especially in three space dimensions or for problems that need the resolution of different scales.

In order to analyze various engineering problems with linear or nonlinear geometries, a stable and efficient numerical method is of great importance. Also, it is essential to develop a powerful algorithm appropriate for a wide range of problems. Dynamic relaxation (DR) method has been proved to have a promising potential with a number of distinguished features. For instance, it has a clear and simple algorithm so that the required computer programming is straightforward. Moreover, it needs to solve large scale equations directly, because of its explicit formulation. Also, it is very reliable and stable to solve and analyze nonlinear problem.

The DR technique is based on the fact that a system undergoing damped vibration ultimately comes to rest in the displaced position of the static equilibrium. The damped vibration starts when exciting the system by a constant force. The method can be interpreted both by physics and mathematics. Physically, the procedure is similar to obtaining the steady state solution of a dynamic system. Accordingly, in order to achieve the solution of a static problem, it could be transferred to a fictitious dynamic space. Due to this transference, it is necessary to specify some extra factors for the problem. These factors are mass, damping

coefficient and time step. Mathematically, the DR method can also be generated from the second order Richardson rule. The convergence acceleration could be investigated in a pure mathematical method.

The dynamic relaxation (DR) is a numerical method usually used in the form finding of all kind of structures (tensegrity structures, membrane structures, shell structures ...etc.) that consists in considering that the mass of the system is discretized and lumped in the nodes; these nodes oscillate about the equilibrium position, and by introducing artificial inertia and damping elements, the nodes come to rest in the static equilibrium position. The fact of using artificial inertia and damping makes the use of DR methods be restrained to the cases where the only objective of the calculation is to obtain the final equilibrium position of the structure, because the transient part will not be physical. However, the displacement path is close to a physical one, as it will be shown in this work. In the literature, we can find DR methods using kinetic damping and DR methods using viscous damping. The methodology of these two methods is different. In the case of DR with kinetic damping, the kinetic energy of the structure is traced, and the velocities are reset to zero at each of the kinetic energy peaks (i.e. that are gradually smaller) until the balance of internal and external forces is reached and the structure comes to rest; therefore, the principle in this case is to try to optimize the mass matrix in order to reach as fast as possible the kinetic energy peaks. On the other hand, when using DR with viscous damping, the velocities are not so important; the main idea is to try to damp as effectively as possible the oscillations, by searching an optimum viscous damping coefficient. The most commonly used damping method is the viscous damping. This method is closer to the real behavior of the structures, since they behave as if they were somehow viscous. The kinetic damping makes the structure evolve in a very different way.

Dynamic Relaxation method (DR) Coupled with Finite Differences method (FD) is used for solving ordinary and partial differential equations as a single equation or as a group of differential equations. To apply dynamic relaxation

software technique, the differential equations are transformed into dynamic equations by adding damping and inertia elements. These in turn are expressed in finite differences form, and the solution is obtained by an iterative procedure as is explained in the following paragraphs:

The differential equation is referred to in the following as:

$$f = 0 \quad (1.1)$$

Where, $f = 0$, may be an ordinary differential equation as follows:

$$P(x) \frac{d^2u}{dx^2} + Q(x) \frac{du}{dx} + R(s)u = 0$$

Or a partial differential equation as stated below:

$$P(x) \frac{\partial^2u}{\partial x^2} + Q(x) \frac{\partial^2u}{\partial y^2} + R(x, y)u = 0$$

1.2 Formulation of Dynamic Relaxation Equations:

The dynamic relaxation method (DR) formula begins with the dynamic equation which may be written as:

$$f = \rho \frac{\partial^2u}{\partial t^2} + k \frac{\partial u}{\partial t} \quad (1.2)$$

In this procedure the statically differential system i.e. equation (1.1) is transferred to an artificial dynamic space by adding fictitious inertia and damping forces as in equation (1.2).

The DR method was first proposed in 1960s; refer to Rushton [1], Cassel and Hobbs [2], and Day [3]. In this method, the equations of equilibrium are converted to dynamic equations by adding damping and inertia terms, these are then expressed in finite difference form and solution is obtained through iterations. The optimum damping coefficient and time increment used to stabilize the solution depend on a

number of factors including the stiffness matrix of the structure, the applied load, the boundary conditions and the size of the mesh used, etc.

In order to analyze various complicated problems in engineering, many kinds of efficient numerical methods such as finite difference method, finite element method and the weighted residual method have been developed. However, the accompanying problem is that large computers are needed to solve the related large scale equations. Sometimes, the equations are so large that one can only obtain rough results. This is especially conspicuous in solving non – linear problems. In addition, numerical instability during iteration is often involved.

In the traditional methods of solving equations from static equilibrium problems, it is considered that internal forces exist initially in the structures. In so doing, one assumes that the external forces were exerted very slowly so that the dynamic process of the structures could be neglected. In fact, as has been pointed out by Rayleigh [4], static solution of a mechanics system can be referred to as the steady state part of the transient response of the system to step loading. This approach was successfully applied to solving linear problems by Otter [5] and Day [3] independently in 1965, and was named the dynamic relaxation (DR) method.

Nowadays, researchers are attracted by the efficiency of solving non – linear problems with DR. The applications of DR to various problems indicate that the method has the following distinctive features { see, for example, [6] – [9] }.

Numerical techniques other than the dynamic relaxation (DR) method include finite element method (FEM), which is widely used in most of the theoretical analyses of today's research. In a comparison between the dynamic relaxation method and the finite element method, Aalami [10] found that the computer time required for finite element method is eight times greater than that for the dynamic relaxation analysis, whereas storage capacity for finite element analysis is ten times or more than that for DR analysis. This fact is supported by Putcha and Reddy [11], and Turvey and Osman{ [12] – [14]}, who they noted that some of the finite element

formulations require large storage capacity and computer time. However, if the analysis requires less computations and computer time, then, the dynamic relaxation is considered more efficient than the finite element method. In another comparison Aalami [10] found that the difference in accuracy between one version of finite element and another may reach a value of 10% or more, whereas a comparison between one version of finite element method and DR showed a difference of more than 15%. Therefore, the dynamic relaxation method (DR) can be considered of acceptable accuracy.

The only apparent limitation of dynamic relaxation (DR) method is that it can only be applied to limited geometries. However, this limitation is irrelevant to square and rectangular plates and beams which are widely used in engineering applications.

The errors inherent in the dynamic relaxation (DR) technique {[15] – [23]} include discretization error which is due to the replacement of a continuous function with a discrete function. Also, there is an additional error resulting from the non-exact solution of the discrete equations due to the variations of the velocities from the edges of the plate to the center. The usage of finer meshes reduces the discretization error, but increases the round – off error due to the large amount of computations involved.

For the sake of simplifying and explanation of the DR method, u in equation (1.2) is referred to as displacement, and hence the terms $\partial u / \partial t$ and $\partial^2 u / \partial t^2$ are the velocity and acceleration respectively. Accordingly the first and second terms on the right – hand side are the inertia and damping terms respectively. ρ And k are the inertia and damping coefficients respectively, and t is time.

If the velocities before and after the period Δt at an arbitrary node in the finite difference mesh are denoted by $\{\partial u / \partial t\}_{n-1}$ and $\{\partial u / \partial t\}_n$ respectively, then using finite differences in time, and specifying the value of the function at $(n - \frac{1}{2})$, it is possible to write equation (1.2) in the following form:

$$f_{n-\frac{1}{2}} = \frac{\rho}{\Delta t} \left[\left\{ \frac{\partial u}{\partial t} \right\}_n - \left\{ \frac{\partial u}{\partial t} \right\}_{n-1} \right] + k \left\{ \frac{\partial u}{\partial t} \right\}_{n-\frac{1}{2}} \quad (1.3)$$

Now $\left\{ \frac{\partial u}{\partial t} \right\}_{n-\frac{1}{2}}$, which is the velocity at the middle of the time increment, can be approximated by the mean velocities before and after the time increment, Δt , which is expressed as follows:

$$\left\{ \frac{\partial u}{\partial t} \right\}_{n-\frac{1}{2}} = \frac{1}{2} \left[\left\{ \frac{\partial u}{\partial t} \right\}_n - \left\{ \frac{\partial u}{\partial t} \right\}_{n-1} \right]$$

Hence, equation (1.3) can be expressed in the following form as:

$$f_{n-\frac{1}{2}} = \frac{\rho}{\Delta t} \left[\left\{ \frac{\partial u}{\partial t} \right\}_n - \left\{ \frac{\partial u}{\partial t} \right\}_{n-1} \right] + \frac{k}{2} \left[\left\{ \frac{\partial u}{\partial t} \right\}_n - \left\{ \frac{\partial u}{\partial t} \right\}_{n-1} \right] \quad (1.4)$$

Equation (1.4) can then be arranged to give the velocity after the time interval, Δt :

$$\left\{ \frac{\partial u}{\partial t} \right\}_n = (1 + k^*)^{-1} \left[\frac{\Delta t}{e} f_{n-\frac{1}{2}} + (1 - k^*) \left\{ \frac{\partial u}{\partial t} \right\}_{n-1} \right] \quad (1.5)$$

Where:

$$k^* = \frac{k\Delta t}{2\rho}$$

The displacements at the middle of the next time increment can be determined by integrating the velocity, so that:

$$u_{n+\frac{1}{2}} = u_{n-\frac{1}{2}} + \left\{ \frac{\partial u}{\partial t} \right\}_n \Delta t \quad (1.6)$$

The iterative procedure begins at time $t = 0$ with all initial values of the velocities and displacements equal to zero or any other suitable values. In the first iteration, the velocities are obtained from equation (1.5) and the displacements from

equation (1.6). The boundary conditions are then applied. Subsequent iterations follow the same steps until the desired accuracy is achieved.

1.3 Finite Difference Approximation:

1.3.1 Ordinary Differential Equations:

The values of the interpolating function $u(x)$ in the vicinity of the node i in a non – uniform or graded mesh shown in figure (1.1) below can be expressed as follows using Taylor's series:

$$u(i + 1) = u(i) + P_{i+1}\Delta x u'(i) + \frac{P_{i+1}^2\Delta x^2}{2!}u''(i) + \frac{P_{i+1}^3\Delta x^3}{3!}u'''(i) + \frac{P_{i+1}^4\Delta x^4}{4!}u''''(i) + \dots \quad (1.7)$$

$$u(i - 1) = u(i) - P_i\Delta x u'(i) + \frac{P_i^2\Delta x^2}{2!}u''(i) - \frac{P_i^3\Delta x^3}{3!}u'''(i) + \frac{P_i^4\Delta x^4}{4!}u''''(i) + \dots \quad (1.8)$$

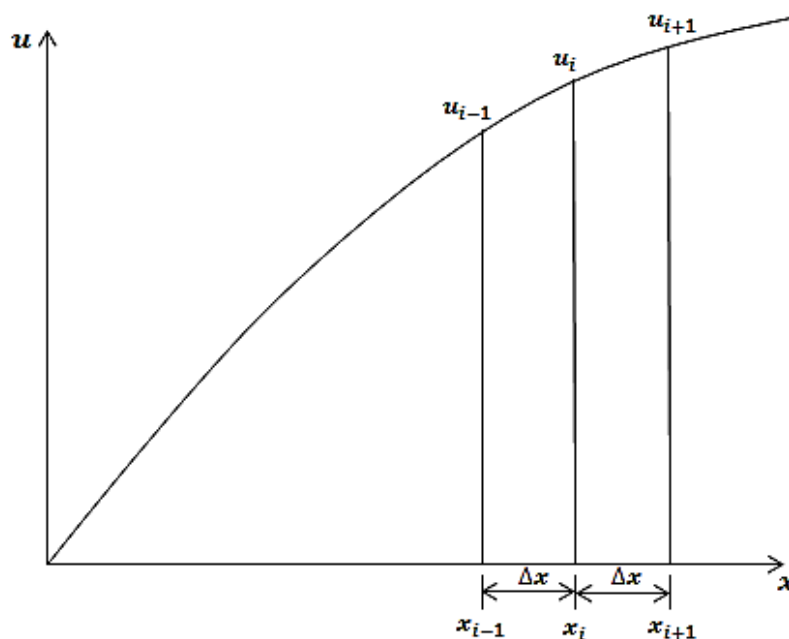


Figure (1.1) Non – uniform or graded mesh

Where $u'(i), u''(i), u'''(i)$, and $u''''(i)$ are the first, second, third, and fourth derivatives of the function $u(x)$ at node i .

When multiplying equation (1.7) by P_i^2 and equation (1.8) by P_{i+1}^2 , then subtract the latter from the former and rearrange the resulting expression to obtain the function at node i as follows:

$$u'(i) = \frac{1}{\Delta x} \left[\alpha_i^{(1)} u(i+1) + \alpha_i^{(2)} u(i) + \alpha_i^{(3)} u(i-1) \right] + \epsilon_1 \quad (1.9)$$

Where:

$$\alpha_i^{(1)} = \frac{p_i}{P_{i+1}(P_i + P_{i+1})}$$

$$\alpha_i^{(2)} = -\frac{p_i - p_{i+1}}{P_i P_{i+1}}$$

$$\alpha_i^{(3)} = -\frac{P_{i+1}}{p_i(p_i + P_{i+1})}$$

$$\epsilon_1 = -p_i p_{i+1} \frac{\Delta x^2}{6} u'''''' + \dots \quad (1.10)$$

Multiply equation (1.7) by p_i and equation (1.8) by p_{i+1} and add them together to obtain the second derivative of the function $u(x)$ at node i as follows:

$$u''(i) = \frac{2}{\Delta x^2} \left[\alpha_i^{(4)} u(i+1) + \alpha_i^{(5)} u(i) + \alpha_i^{(6)} u(i-2) \right] + \epsilon_2 \quad (1.11)$$

Where:

$$\alpha_i^{(4)} = \frac{1}{P_{i+1}(p_i + p_{i+1})}$$

$$\alpha_i^{(5)} = -\frac{1}{P_i p_{i+1}}$$

$$\alpha_i^{(6)} = \frac{1}{p_i(p_i + p_{i+1})}$$

$$\epsilon_2 = \frac{P_i - P_{i+1}}{3} \Delta x u'''_i - \frac{p_i^2 - p_i p_{i-1} + p_{i+1}}{12} \Delta x^2 u''''_i + \dots \quad (1.12)$$

If the derivatives of $f(x)$ which are greater than the third are assumed negligible i.e. the actual function approximates a quadratic function then ϵ_1 and ϵ_2 represent the error in the approximation resulting from replacing the actual function by a quadratic function. The error in the first derivative of the function of equation (1.10) depends on the graded mesh and it is proportional to Δx^2 . The error in the second derivative of the function, equation (1.12), is proportional to Δx^2 for a uniform mesh (i.e. $P_i = P_{i+1}$), and proportional to Δx for a graded mesh (i.e. $P_i \neq P_{i+1}$). That is to say the error associated with a graded mesh is greater than that of a uniform mesh with the same number of elements. However, a graded mesh is more flexible than a uniform mesh and it allows closer nodes to be employed in those regions where a higher degree of accuracy is required.

When the mesh is uniform $P_i = P_{i+1}$, and hence:

$$\alpha_i^{(1)} = \alpha_i^{(4)} = \frac{1}{2}$$

$$\alpha_i^{(1)} = 0$$

$$\alpha_i^{(1)} = \alpha_i^{(6)} = -\frac{1}{2}$$

$$\alpha_i^{(5)} = -1$$

And therefore, the first and second derivatives, with the error neglected, are as follows:

$$\frac{du}{dx}(i) = \frac{1}{2\Delta x} [u(i+1) - u(i-1)] \quad (1.13)$$

$$\frac{d^2u}{dx^2}(i) = \frac{1}{\Delta x^2} [u(i+1) - 2u(i) + u(i-1)] \quad (1.14)$$

The first derivative of the function with respect to x can be written also for a uniform mesh as follows with:

$$\frac{du}{dx}(i) = \frac{1}{\Delta x} [u(i) - u(i-1)]$$

Or

$$\frac{du}{dx}(i) = \frac{1}{\Delta x} [u(i+1) - u(i)] \quad (1.15)$$

1.3.2 Partial Differential Equations:

The first and second derivatives of a function $u(x, y)$ at an arbitrary node (i, j) shown in figure (1.2) below can be written as follows:

$$\frac{\partial u}{\partial x}(i, j) = \frac{1}{\Delta x} [\alpha_{ij}^{(1)} u(i+1, j) + \alpha_{ij}^{(2)} u(i, j) + \alpha_{ij}^{(3)} u(i-1, j)]$$

$$\frac{\partial^2 u}{\partial x^2}(i, j) = \frac{2}{\Delta x^2} [\alpha_{ij}^{(4)} u(i+1, j) + \alpha_{ij}^{(5)} u(i, j) + \alpha_{ij}^{(6)} u(i-1, j)]$$

$$\frac{\partial u}{\partial x \partial y}(i, j) = \frac{1}{\Delta x \Delta y} [\alpha_{ij}^{(1)} \beta_{ij}^{(1)} u(i+1, j+1) + \alpha_{ij}^{(1)} \beta_{ij} u(i+1, j-1)$$

$$+ \alpha_{ij}^{(3)} \beta_{ij}^{(1)} u(i-1, j+1) + \alpha_{ij}^{(3)} \beta_{ij}^{(3)} u(i-1, j-1)]$$

$$\frac{\partial u}{\partial y}(i, j) = \frac{1}{\Delta y} [\beta_{ij}^{(1)} u(i, j+1) + \beta_{ij}^{(2)} u(i, j) + \beta_{ij}^{(3)} u(i, j-1)]$$

$$\frac{\partial^2 u}{\partial y^2} = \frac{2}{\Delta y^2} [\beta_{ij}^{(4)} u(i, j+1) + \beta_{ij}^{(5)} u(i, j) + \beta_{ij}^{(6)} u(i, j-1)]$$

Where:

$$\alpha_{ij}^{(1)} = \frac{P_i}{P_{i+1}(P_{i+1} + P_i)}$$

$$\alpha_{ij}^{(2)} = \frac{P_{i+1} - P_i}{P_i P_{i+1}}$$

$$\alpha_{ij}^{(3)} = -\frac{P_{i+1}}{P_i(P_i + P_{i+1})}$$

$$\alpha_{ij}^{(4)} = \frac{1}{P_i + P_{i+1}}$$

$$\alpha_{ij}^{(5)} = -\frac{1}{P_i P_{i+1}}$$

$$\alpha_{ij}^{(6)} = \frac{1}{P_i(p_i + P_{i+1})}$$

$$\beta_{ij}^{(1)} = \frac{r_i}{r_{i+1}(r_{i+1} + r_i)}$$

$$\beta_{ij}^{(2)} = \frac{r_{i+1} - r_i}{r_i r_{i+1}}$$

$$\beta_{ij}^{(3)} = -\frac{r_{i+1}}{r_i(r_i + r_{i+1})}$$

$$\beta_{ij}^{(4)} = \frac{1}{r_i + r_{i+1}}$$

$$\beta_{ij}^{(5)} = -\frac{1}{r_i r_{i+1}}$$

$$\beta_{ij}^{(6)} = \frac{1}{r_i(r_i + r_{i+1})}$$

Where P_i and P_{i+1} are the ratios of the dimensions of the elements on both sides of the node (i, j) to the average element length all measured in the x – direction. r_i And

r_{i+1} are the ratios of the dimensions of the elements on both sides of node (i, j) to the average element length all measured in the y – direction.

	$i - 1, j + 1$	$i, j + 1$	$i + 1, j + 1$
$r_{j+1}\Delta y$	$i - 1, j$	i, j	$i + 1, j$
$r_j\Delta y$	$i - 1, j - 1$	$i, j - 1$	$i + 1, j - 1$
	$P_i\Delta x$	$P_{i+1}\Delta x$	

Figure (1.2) the first and second derivatives of a two dimensional function $u(x, y)$

When the mesh is uniform i.e. $p_i = P_{i+1}$ And $r_i = r_{i+1}$, we have:

$$\alpha_{ij}^{(1)} = \beta_{ij}^{(1)} = \frac{1}{2}$$

$$\alpha_{ij}^{(2)} = \beta_{ij}^{(2)} = 0$$

$$\alpha_{ij}^{(3)} = \beta_{ij}^{(3)} = -\frac{1}{2}$$

$$\alpha_{ij}^{(4)} = \beta_{ij}^{(4)} = \frac{1}{2}$$

$$\alpha_{ij}^{(5)} = \beta_{ij}^{(5)} = -1$$

$$\alpha_{ij}^{(6)} = \beta_{ij}^{(6)} = \frac{1}{2}$$

The first and second derivatives of $u(x, y)$ for a uniform mesh are:

$$\frac{\partial u}{\partial x}(i, j) = \frac{1}{2\Delta x} [u(i+1, j) - u(i-1, j)] \quad (1.16)$$

$$\frac{\partial u}{\partial y}(i, j) = \frac{1}{2\Delta y} [u(i, j+1) - u(i, j-1)] \quad (1.17)$$

$$\frac{\partial^2 u}{\partial x^2}(i, j) = \frac{1}{\Delta x^2} [u(i+1, j) - 2u(i, j) + u(i-1, j)] \quad (1.18)$$

$$\frac{\partial^2 u}{\partial y^2}(i, j) = \frac{1}{\Delta y^2} [u(i, j+1) - 2u(i, j) + u(i, j-1)] \quad (1.19)$$

$$\begin{aligned} \frac{\partial^2 u}{\partial x \partial y}(i, j) &= \frac{1}{4\Delta x \Delta y} [u(i+1, j+1) - u(i+1, j-1) \\ &\quad - u(i-1, j+1) + u(i-1, j-1)] \end{aligned} \quad (1.20)$$

CHAPTER TWO

Procedural Steps in Solving Differential

Equations Using DR Method

2.1 The Dynamic Relaxation (DR) Program:

The DR program performs the following operations:

1. Reads data file.
2. Computes fictitious densities.
3. Computes velocities and displacements.
4. Checks stability of numerical computations.
5. Checks convergence of solution.
6. Checks wrong convergence.

Refer to references {[24] – [30]} for more information about analysis of rectangular laminated plates in bending.

2.1.1 Numerical Instability:

In every iteration, the value of the function at the center of the solution domain or other suitable point is compared with two estimated reference values representing lower and upper bounds of the function at that point. If solution was failed such that the computed value of the function at the specified point did not fall within the prescribed range, the solution is deemed unstable, and therefore iterations are terminated. The damping coefficients are then reduced and the process of iteration is restarted once again. The iterations are repeated several times until stability is reached.

2.1.2 Convergence of DR Solution:

Convergence of the dynamic relaxation solution is checked at the end of each iteration by comparing the velocities over the domain with a prescribed value. The procedure is repeated until the solution is deemed converged and consequently the iterative process is terminated.

2.1.3 Convergence to an Invalid Solution:

Sometimes DR solution converges to incorrect answer. Check for invalid solution is carried out after the solution has satisfied the convergence criterion explained earlier. In the check procedure the profile of variable is compared with the anticipated profile over the domain. For instance, if the value of the function on the boundaries is zero, and it is known that the function increases from edge to center, and then the solution should follow a similar profile. If the computed profile is different from that, the solution is deemed to be incorrect. When this happens, the solution can hardly be made to converge to the correct answer by altering the damping coefficients and time increment. One should take another look to the boundary conditions and correct them if they are wrong.

2.1.4 Time Increment:

Proper time increment is a very important factor for speeding convergence and controlling numerical computations. When time increment is too small, convergence becomes tediously slow; and if it is too large, the solution becomes unstable. Time increment must be less than 1, say, 0.8.

2.1.5 Damping Coefficient:

The optimum damping coefficient is that which produces critical motion. When the damping coefficient or coefficients are large, the motion is over – damped and convergence becomes very slow. When the coefficients are small, the motion is under – damped and can cause numerical instability.

2.2 Solved Examples:

In the following examples, the dynamic relaxation (DR) numerical method combined with the finite differences discretization technique is used to solve nonlinear ordinary and partial differential equations. Subsequently a FORTRAN program is developed to generate the numerical results as analytical and/ or exact solutions.

2.2.1 Solution of Ordinary Differential Equation:

Example (1):

Solve the following ordinary differential equation using the dynamic relaxation (DR) method.

$$\frac{d^2w}{dx^2} + q = 0 \quad (2.1)$$

Where $q = \pi^2 \sin(\pi x)$, and the end conditions are:

$$w(0) = w(1) = 0$$

Note that the exact solution is: $w = \sin(\pi x)$.

Solution:

Write equation (2.1) in finite difference form as shown below:

$$f = \frac{1}{\Delta x^2} [w(i+1) - 2w(i) + w(i-1)] + q(i)$$

The velocities are:

$$w_t(i)_n = \frac{1}{1 + k^*(i)} \left[1 - k^*(i)w_t(i)_{n-1} + \frac{f_{n-\frac{1}{2}} \Delta t}{\rho(i)} \right]$$

The values of the function are computed from:

$$w(i)_{n+\frac{1}{2}} = w(i)_{n-\frac{1}{2}} + w_t(i)_n \Delta t$$

Now if the region of the problem $\{0 - 1\}$ is divided into 10 elements, then the end conditions can be expressed as:

$$w(0) = w(10) = 0$$

However, in this case and due to symmetry of end conditions, the solution can be obtained over half the domain {i.e. $0 - 1/2$ }. The condition at the symmetry line defined by $i = 5$ is:

$$w(6) = w(4)$$

All initial values are set to zero and iterations are started. After each iteration the velocities are compared with a reference of very small value of about 10^{-6} . When all velocities are less than the prescribed value, the process is terminated. The process may be terminated of course when the maximum value of the function (i.e. at center) exceeds certain bounds which indicate that the solution is becoming unstable. These bounds are defined by the inequalities $0 \leq w \leq 5$. After the solution has converged, a further check is made to guarantee that the solution has converged correctly. To facilitate this use is made of the fact that function increases from end to center. In fact this profile is achieved by the converged solution and therefore the solution is considered to be correct.

The computer output is listed in table (2.1) below. The solution was converged in 136 iterations. Note the close comparison between the approximate and exact solutions.

Table (2.1) solution of example (1)

x	0.10	0.20	0.30	0.40	0.50
w (approximate solution)	0.3119	0.5932	0.8164	0.9598	1.0091
w (exact solution)	0.3091	0.5880	0.8092	0.9512	1.0000

The FORTRAN program entitled [Osama 1. FOR] is shown below which is used to solve an ordinary differential equation of example (2.1) using DR method.

Osama 1. FOR

C This program solves an ordinary differential equation using DR

C method

Real K, KS

Dimension X (0:10), Q (0:10), W (0:10), WT (0:10)

Open (unit=5, File='Osama 1. Dat', status='old')

Open (unit = 6, File = 'Osama 1. Out', status = 'unknown')

Read (5, *) K, DT, RHO, NMAX

8 Format (1x, 'x', 8x, 'w')

9 Format (F4.2, 5x, F6.4)

Dx = 0.1

$$PI = 3.1416$$

$$KS = K * DT / (2.0 * RHO)$$

$$DO 10I = 0, 10$$

$$X (I) = Dx * I$$

$$Q (I) = (PI ** 2) * \sin (PI * X (I))$$

10 Continue

C Initial Conditions

$$DO 20I = 0, 10$$

$$WT (I) = 0.0$$

$$W (I) = 0.0$$

20 Continue

$$DO 30N = 1, NMax$$

$$DO 40 I = 1, 9$$

$$F = ((W (I+1) - 2.0 * W (I) + W (I - 1)) / (Dx ** 2)) + Q (I)$$

$$WT (I) = ((1.0 - KS) * WT (I) + F * DT / RHO) / (1.0 + KS)$$

40 Continue

$$DO 50 I = 1, 9$$

$$W (I) = W (I) + WT (I) * DT$$

50 Continue

C Boundary Conditions

W (0) = 0.0

W (10) = 0.0

C Check Instability

NL = 0

IF (W (5). GT. 5.0) Then

NL = 1

Go To 61

Else

Continue

End if

C Convergence Criterion

LL = 0

Do 31 I = 0, 10

IF (WT (I). GT. 0.000001. OR. WT (I). LT. 0.000001) LL = LL + 1

31 Continue

IF (LL. GT. 0) Then

Go To 30

Else

Go To 100

End if

30 Continue

100 Write (6, *) 'Number of iterations = ', NMAX

Write (6, 8)

Do 60 I = 0, 10

Write (6, 9) X (I), W (I)

60 Continue

61 IF (NL. EQ. 1) WRITE (6, *) 'Numerical instability is experienced'

Stop

End

The FORTRAN program entitled Osama 2. FOR which is shown below is used to solve ordinary equation of example (2.1) using exact solution method.

Osama 2. FOR

C This program solves an ordinary differential equation using exact solution

C method

Dimension W (0:10), X (0:10)

Dx = 0.1

PI = 3.1416

Do 10 I = 0, 10

X (I) = DX * I

10 Continue

Do 20 I = 1, 9

W (I) = sin (PI * X (I))

20 Continue

Write (6, 8)

8 Format (1x, 'X', 8x, 'W')

Do 30 I = 0, 10

Write (6, 9) X (I), W (I)

9 Format (1x, F4.2, 5x, F6.4)

30 Continue

Stop

End

2.2.2 Solution of Partial Differential Equations:

Example (2.2):

Using the dynamic relaxation (DR) method try to solve the following partial differential equation:

$$\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} + 2\pi^2 u = 0 \quad (2.2)$$

Subject to boundary conditions: $u(x, 0) = 0, u(x, 0.5) = \sin(\pi x), u(0, y) = 0, u(0.5, y) = \sin(\pi y)$

The exact solution is as follows:

$$u = \sin(\pi x) \sin(\pi y)$$

The Computer output is listed in table (2.2) below. The first row of each set is the approximate solution whereas the second row is the exact value. The solution of this example was converged in 73 iterations.

Solution:

Table (2.2) Solution of example (2.2)

0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
0.0000	0.0961	0.1826	0.2511	0.2948	0.3091
0.0000	0.0956	0.1818	0.2502	0.2941	0.3091
0.0000	0.1826	0.3472	0.4775	0.5606	0.5880
0.0000	0.1818	0.3457	0.4758	0.5593	0.5880
0.0000	0.2511	0.4775	0.6568	0.7712	0.8092
0.0000	0.2502	0.4758	0.6549	0.7698	0.8092
0.0000	0.2948	0.5606	0.7712	0.9060	0.9512
0.0000	0.2941	0.5593	0.7698	0.9048	0.9512

0.0000	0.3091	0.5880	0.8092	0.9512	1.0000
0.0000	0.3091	0.5880	0.8092	0.9512	1.0000

Equation (2.2) is written in finite differences form as follows:

$$\begin{aligned}
 f = & \frac{1}{\Delta x^2} [u(i+1, j) - 2u(i, j) + u(i-1, j)] \\
 & + \frac{1}{\Delta y^2} [u(i, j+1) - 2u(i, j) + u(i, j-1)] \\
 & + 2\pi^2 u
 \end{aligned}$$

The FORTRAN program entitled Osama 3. FOR which is shown below is used to solve ordinary differential equation of example (2.2) using DR method.

Osama 3. FOR

C This program solves a partial differential equation using DR method

Real K, KS

Dimension X (0:10), Y (0:10), U (0:10, 0:10), UT (0:10, 0:10)

Open (unit = 5, File = 'Osama 3. Dat', status = 'old')

Open (unit = 6, File = 'Osama 3. Out', status = 'unknown')

Read (5, *) K, DT, RHO, NMax

8 Format (2x, 'U (I, J)')

9 Format (6 (2x, F 6.4))

$$NL = 0$$

$$DX = 0.1$$

$$DY = 0.1$$

$$PI = 3.1416$$

$$KS = K * DT / (2.0 * RHO)$$

$$\text{Do } 10 \text{ I} = 0, 5$$

$$X(I) = DX * I$$

10 Continue

$$\text{Do } 11 \text{ J} = 0, 5$$

$$Y(J) = DY * J$$

11 Continue

$$\text{Do } 20 \text{ I} = 0, 5$$

$$\text{Do } 20 \text{ J} = 0, 5$$

$$U(I, J) = 0.0$$

$$UT(I, J) = 0.0$$

20 Continue

$$\text{Do } 30 \text{ N} = 1, \text{NMax}$$

$$\text{Do } 40 \text{ I} = 1, 4$$

$$\text{Do } 40 \text{ J} = 1, 4$$

$$F = (U(I+1, J) - 2.0 * U(I, J) + U(I-1, J)) / (DX ** 2) + (U(I, J+1)$$

$$- 2.0 * U (I, J) + U (I, J-1)) / (DY ** 2) + 2.0 * PI ** 2 * U (I, J)$$

$$UT (I, J) = ((1.0 - KS) * KS) * UT (I, J) + F * DT / RHO) / (1.0 + KS)$$

40 Continue

Do 50 I = 1, 4

Do 50 J = 1, 4

$$U (I, J) = U (I, J) + UT (I, J) * DT$$

50 Continue

C Boundary conditions

Do 41 I = 0, 5

$$U (I, 0) = 0.0$$

$$U (I, 5) = \sin (PI * X (I))$$

41 Continue

Do 42 J = 0, 5

$$U (0, J) = 0.0$$

$$U (5, J) = \sin (PI * Y (J))$$

42 Continue

C Check Instability

IF (U (5, 5). GT. 5.0) Then

NL = 1

Go To 61

Else

Continue

End if

C Convergence Criterion

LL = 0

Do 31 I = 0, 5

Do 31 J = 0, 5

IF (UT (I, J). LT. 0.000001) LL = LL + 1

31 Continue

If (LL. GT. 0) Then

Go To 30

Else

Go To 100

End if

30 Continue

100 Write (6, *) 'Number of iterations = ', NMax

Write (6, 8)

Do 60 I = 0, 5

Write (6, 9) (U (I, J), J = 0.5)

60 Continue

61 If (NL. EQ. 1) write (6, *) 'Numerical instability is experienced'

Stop

End

The FORTRAN program entitled Osama 4. FOR is illustrated below which is used to solve a partial differential equation of example (2.2) using the exact or analytical solution.

Osama 4. FOR

C This program solves an ordinary differential equation

C using exact solution method

Dimension U (0:10, 0:10), X (0:10), Y (0:10)

DX = 0.1

DY = 0.1

PI = 3.1416

Do 10 I = 0, 10

Do 20 J = 0, 10

X (I) = DX * I

Y (J) = DY * J

10 Continue

Do 20 I = 1, 9

$$U(I, J) = \sin(\pi * X(I)) * \sin(\pi * Y(J))$$

20 Continue

Do 30 I = 0, 5

WRITE (6, 9) (U(I, J), J = 0, 5)

9 Format (6 (1 X, F 6.4))

30 Continue

Stop

END

Example (2.3):

Solve the following system of partial differential equations over a square domain bounded by $0 \leq x \leq 1$ and $0 \leq y \leq 1$ using DR method.

$$\left. \begin{aligned} \frac{\partial^2 u}{\partial x^2} + 2 \frac{\partial^2 v}{\partial x \partial y} + \frac{\partial^2 u}{\partial y^2} - u + \frac{\partial w}{\partial x} &= 0 \\ \frac{\partial^2 v}{\partial x^2} + 2 \frac{\partial^2 u}{\partial x \partial y} + \frac{\partial^2 v}{\partial y^2} - v + \frac{\partial w}{\partial y} &= 0 \\ \frac{\partial^2 w}{\partial x^2} + \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial^2 w}{\partial y^2} + q &= 0 \end{aligned} \right\} \quad (2.3)$$

Where $q = \pi \left(\pi^2 + \frac{1}{2} \right) \sin(\pi x) \sin(\pi y)$

Solution:

The boundary conditions are as given in the next figure (i.e. figure (2.3)).

Note that the differential equations are satisfied by the following solutions:

$$u = \frac{1}{8} \cos(\pi x) \sin(\pi y)$$

$$v = \frac{1}{8} \sin(\pi x) \cos(\pi y)$$

$$w = \frac{1 + 4\pi^2}{8\pi} \sin(\pi x) \sin(\pi y)$$

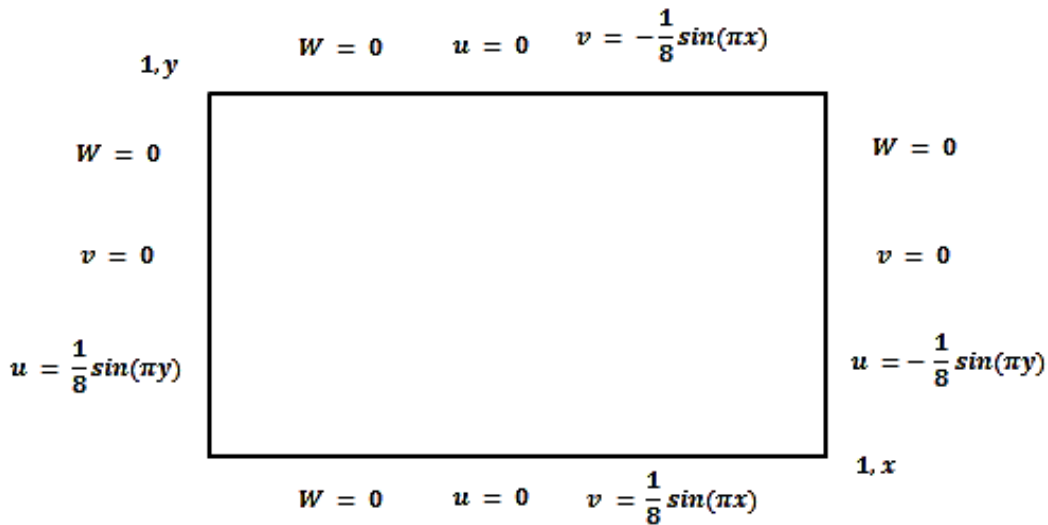


Figure (2.3) Boundary Conditions of Example (2.3)

Write equation (2.3) in finite difference form:

$$f_1 = \frac{1}{\Delta x^2} [u(i+1, j) - 2u(i, j) + u(i-1, j)]$$

$$+ \frac{1}{4\Delta x \Delta y} [v(i+1, j+1) - v(i+1, j-1) - v(i-1, j+1) + v(i-1, j-1)]$$

$$+ \frac{1}{\Delta y^2} [u(i, j+1) - 2u(i, j) + u(i, j-1)] - u(i, j)$$

$$\begin{aligned}
& + \frac{1}{2\Delta x} [w(i+1, j) - w(i-1, j)] \\
f_2 &= \frac{1}{\Delta x^2} [v(i+1, j) - 2v(i, j) + v(i-1, j)] \\
& + \frac{1}{4\Delta x \Delta y} [u(i+1, j+1) - u(i+1, j-1) - u(i-1, j+1) + u(i-1, j-1)] \\
& + \frac{1}{\Delta y^2} [v(i, j+1) - 2v(i, j) + v(i, j-1)] - v(i, j) \\
& + \frac{1}{2\Delta y} [w(i, j+1) - w(i, j-1)] \\
f_3 &= \frac{1}{\Delta x^2} [w(i+1, j) - 2w(i, j) + w(i-1, j)] \\
& + \frac{1}{2\Delta x} [u(i+1, j) - u(i-1, j)] + \frac{1}{2\Delta y} [v(i, j+1) - v(i, j-1)] \\
& + \frac{1}{\Delta y^2} [w(i, j+1) - 2w(i, j) + w(i, j-1)] + q(i, j)
\end{aligned}$$

Compute the velocities:

$$u_t(i, j)_n = \frac{1}{1 + k_u^*(i, j)} \left\{ [1 - k_u^*(i, j)] u_t(i, j)_{n-\frac{1}{2}} + \left(\frac{f_1 \Delta t}{\rho_u(i, j)} \right)_{n+\frac{1}{2}} \right\}$$

$$v_t(i, j)_n = \frac{1}{1 + k_v^*(i, j)} \left\{ [1 - k_v^*(i, j)] v_t(i, j)_{n-\frac{1}{2}} + \left(\frac{f_2 \Delta t}{\rho_v(i, j)} \right)_{n+\frac{1}{2}} \right\}$$

$$w_t(i, j)_n = \frac{1}{1 + k_w^*(i, j)} \left\{ [1 - k_w^*(i, j)] w_t(i, j)_{n-\frac{1}{2}} + \left(\frac{f_3 \Delta t}{\rho_w(i, j)} \right)_{n+\frac{1}{2}} \right\}$$

Compute the values of $u(i, j)$, $v(i, j)$ and $w(i, j)$, from the following equations:

$$u(i, j)_{n+\frac{1}{2}} = u(i, j)_{n-\frac{1}{2}} + u_t(i, j)_n \Delta t$$

$$v(i, j)_{n+\frac{1}{2}} = v(i, j)_{n-\frac{1}{2}} + v_t(i, j)_n \Delta t$$

$$w(i, j)_{n+\frac{1}{2}} = w(i, j)_{n-\frac{1}{2}} + w_t(i, j)_n \Delta t$$

Apply the boundary conditions. If the domain is divided into 100 elements: 10 elements in the x – direction and 10 elements in the y – direction, then:

$$w(0, j) = w(10, j) = 0$$

$$u(0, j) = -u(10, j) = \frac{1}{8} \sin \pi(j \Delta y)$$

$$v(0, j) = v(10, j) = 0$$

$$w(i, 0) = w(i, 10) = 0$$

$$u(i, 0) = u(i, 10) = 0$$

$$v(i, 0) = -v(i, 10) = \frac{1}{8} \sin \pi(j \Delta x)$$

After each iteration check is made on the convergence and stability of the solution. The solution is considered converged when the velocities all over the domain is less than 10^{-6} . The criterion for instability is set by taking the bounds on w as: $0 \leq w \leq 5$. When solution has converged a further check is made for convergence to an invalid answer.

One must remember to exploit symmetry if it exists, a facility provided by the computer program listed next. In this example the solution can be obtained over one quarter of the domain using the following boundary conditions:

$$w(6, j) = w(4, j) = 0, v(6, j) = v(4, j) = 0, u(6, j) = -u(4, j) = 0$$

$$w(i, 6) = w(i, 4) = 0, v(i, 6) = v(i, 4) = 0, u(i, 6) = u(i, 4) = 0$$

The FORTRAN program entitled Osama 5. FOR is shown below which is used to solve a system of partial differential equations of example (2.3) using DR method.

Osama 5. FOR

C This program solves a system of partial differential equations

C using DR method

Real K, KS

Dimension X (0:10), Y (0:10), U (0:10, 0:10), V (0:10, 0:10), W (0:10, 0:10),

UT (0:10, 0:10), VT (0:10, 0:10), WT (0:10, 0:10), Q (0:10, 0:10)

Open (unit = 5, File = 'Osama 5. Dat', status = 'old')

Open (unit = 6, File = 'Osama 5. Out', status = 'unknown')

Read (5, *) K, DT, RHO, NMAX

8 Format (1X, 'U (I, J)')

13 Format (6 (1X, F 6.4))

9 Format (1X, 'V (I, J)')

14 Format (6 (1X, F 6.4))

12 Format (1X, 'W (I, J)')

15 Format (6 (1X, 6.4))

NL = 0

$$DX = 0.1$$

$$DY = 0.1$$

$$PI = 3.1416$$

$$KS = K * DT / (2.0 * RHO)$$

$$\text{Do } 10 \text{ I} = 0, 10$$

$$X(I) = DX * I$$

10 Continue

$$\text{Do } 11 \text{ J} = 0, 10$$

$$Y(J) = DY * J$$

11 Continue

$$\text{Do } 20 \text{ I} = 1, 10$$

$$\text{Do } 20 \text{ J} = 1, 10$$

$$Q(I, J) = PI * ((PI ** 2) + 0.5) * \sin(PI * X(I)) * \sin(PI * Y(J))$$

20 Continue

$$\text{Do } 30 \text{ I} = 0, 10$$

$$\text{Do } 30 \text{ J} = 0, 10$$

$$U(I, J) = 0.0$$

$$UT(I, J) = 0.0$$

$$V(I, J) = 0.0$$

$$VT(I, J) = 0.0$$

$$W(I, J) = 0.0$$

$$WT(I, J) = 0.0$$

30 Continue

Do 40 N = 1, NMAX

Do 50 I = 1, 9

Do 50 J = 1, 9

$$F1 = ((U(I+1, J) - 2.0 * U(I, J) + U(I-1, J)) / (DX ** 2)) + \\ ((V(I+1, J+1) - V(I+1, J-1) - V(I-1, J+1) + V(I-1, J-1)) / \\ (4.0 * DX * DY)) + ((U(I, J+1) - 2.0 * U(I, J) + U(I, J-1)) / \\ (DY ** 2)) - U(I, J) + ((W(I+1, J) - W(I-1, J)) / (2.0 * DX))$$

$$F2 = ((V(I+1, J) - 2.0 * V(I, J) + V(I-1, J)) / (DX ** 2)) + \\ ((U(I+1, J+1) - U(I+1, J-1) - U(I-1, J+1) + U(I-1, J-1)) / \\ (4.0 * DX * DY)) + ((V(I, J+1) - 2.0 * V(I, J) + V(I, J-1)) / \\ (DY ** 2)) - V(I, J) + ((W(I, J+1) - W(I, J-1)) / (2.0 * DY))$$

$$F3 = ((W(I+1, J) - 2.0 * W(I, J) + W(I-1, J)) / (DX ** 2)) + \\ ((U(I+1, J) - U(I-1, J)) / (2.0 * DX)) + ((V(I, J+1) - V(I, J-1)) / \\ (2.0 * DY)) + ((W(I, J+1) - 2.0 * W(I, J) + W(I, J-1)) / \\ (DY ** 2)) + Q(I, J)$$

$$UT(I, J) = ((1.0 - KS) * UT(I, J) + F1 * DT / RHO) / (1.0 + KS)$$

$$VT(I, J) = ((1.0 - KS) * VT(I, J) + F2 * DT / RHO) / (1.0 + KS)$$

$$WT(I, J) = ((1.0 - KS) * WT(I, J) + F3 * DT / RHO) / (1.0 + KS)$$

50 Continue

$$\text{Do } 60 \text{ I} = 1, 9$$

$$\text{Do } 60 \text{ J} = 1, 9$$

$$U(I, J) = U(I, J) + UT(I, J) * DT$$

$$V(I, J) = V(I, J) + VT(I, J) * DT$$

$$W(I, J) = W(I, J) + WT(I, J) * DT$$

60 Continue

C Boundary Conditions

$$\text{Do } 41 \text{ J} = 0, 10$$

$$W(0, J) = 0.0$$

$$W(10, J) = 0.0$$

$$U(0, J) = 0.125 * \sin(\text{PI} * Y(J))$$

$$U(10, J) = -0.125 * \sin(\text{PI} * Y(J))$$

$$V(0, J) = 0.0$$

$$V(10, J) = 0.0$$

41 Continue

$$\text{Do } 42 \text{ I} = 0, 10$$

$$W(I, 0) = 0.0$$

$$W(I, 10) = 0.0$$

$$U(I, 0) = 0.0$$

$$U(I, 10) = 0.0$$

$$V(I, 0) = 0.125 * \sin(\text{PI} * X(I))$$

$$V(I, 10) = -0.125 * \sin(\text{PI} * X(J))$$

42 Continue

C Check Instability

IF (W(5, 5). GT. 5.0) Then

NL = 1

Go To 61

Else

Continue

End if

C Convergence Criterion

LL = 0

Do 31 I = 0, 10

Do 31 J = 0, 10

IF (UT(I, J). LT. 0.000001. OR. VT(I, J). LT. 0.000001. OR.

WT(I, J). LT. 0.000001)) LL = LL + 1

31 Continue

IF (LL. GT. 0) Then

Go To 40

Else

Go To 100

End if

40 Continue

100 Write (6, *) ' Number of iterations = ', NMAX

Write (6, 8)

Do 70 J = 0, 5

Write (6, 13) (U (I, J), I = 0, 5)

70 Continue

Write (6, 9)

Do 80 J = 0, 5

Write (6, 14) (V (I, J), I = 0, 5)

80 Continue

Write (6, 12)

Do 90 J = 0, 5

Write (6, 15) (W (I, J), I = 0.5)

90 Continue

61 IF (NL. EQ. 1) write (6, *) ' Numerical instability is experienced'

Stop

END

Osama 6. FOR

The FORTRAN program entitled Osama 6. FOR is shown below which is used to solve a system of partial differential equations of example (2.3) using exact solution.

C This program solves a system of partial differential equations

C using exact solution

Dimension X (0:10), Y (0:10), U (0:10, 0:10), V (0:10, 0:10), W (0:10, 0:10)

7 Format (2X, 'U (I, J)')

8 Format (6 (2X, F 6.4))

13 Format (2X, 'V (1, J)')

9 Format (6 (2X, F 6.4))

14 Format (2X, 'W (I, J)')

12 Format (6 (2X, F 6.4))

DX = 0.1

DY = 0.1

PI = 3.1416

Do 10 I = 0, 5

X (I) = DX * I

10 Continue

Do 11 J = 0,5

Y (J) = DY * J

11 Continue

Do 20 I = 1, 5

Do 20 J = 1, 5

U (I, J) = 0.125 * cos (PI * X (I)) * sin (PI * Y (J))

V (I, J) = 0.125 * sin (PI * Y (I)) * cos (PI * Y (J))

W (I, J) = ((1.0 + 4.0 * PI **2)/ (8.0 * PI)) * sin (PI * X (I)) * sin (PI * Y (J))

20 Continue

Write (6, 7)

Do 30 J = 0, 5

Write (6, 8) (U (I, J), I = 0, 5)

30 Continue

Write (6, 13)

Do 40 J = 0, 5

Write (6, 9) (V (I, J), I = 0, 5)

40 Continue

Write (6, 14)

Do 50 J = 0, 5

Write (6, 12) (W (I, J), I = 0, 5)

Stop

END

CHAPTER THREE

Conclusions

The basics of this book stand on the ordinary and partial differential equations, which value the price of an option by using dynamic relaxation (DR) techniques. The study of partial differential equations in complete generality is a vast undertaking. As almost all of them are not possible to solve analytically we must rely on numerical methods, and the most popular ones are the finite differences methods coupled with dynamic relaxation techniques.

With this book we do not intend to become experts in few hours in order to solve differential equations numerically, but develop both intuition and technical strength required to survive when such a problem needs to be solved.

In comparison with other numerical methods, the dynamic relaxation technique has its own strengths and weaknesses. The advantages of the DR method are that: (a) the method has a simple algorithm so that it will simplify programming ; (b) the formulation is explicit. Therefore, the required memory is less than other techniques ; (c) this method has a high ability in intense nonlinear behaviors. The disadvantages and limitations of the DR methods are summarized as that: (a) in general, the method is unstable and needs some additional conditions to guarantee numerical stability ; (b) the iterations of the method are done in constant load. This causes some issues in limit points ; (c) in nonlinear analysis, with gentle stiffening, the number of iterations is much more in comparison with other techniques.

In the dynamic relaxation technique, the static equations of the differential equations system will be converted to dynamic equations. Then the inertia and damping terms are added to all of these equations. The iterations of the dynamic relaxation technique can then be carried out in the following procedures:

1. Set all initial values of variables to zeros.
2. Compute the velocities.
3. Compute the displacements.
4. Apply suitable boundary conditions for the displacements.
5. Compute the required variables.
6. Apply the appropriate boundary conditions for the required variables.
7. Check if the convergence criterion is satisfied, if it is not repeat the steps from 2 to 6.

A Dynamic Relaxation (DR) program based on finite differences has been developed for the analysis of one dimensional and two dimensional ordinary and partial differential equations. Finite differences coupled with dynamic relaxation method (DR) have been developed. FORTRAN programs have been compiled which they yielded results for a wide range of examples written and solved with the dynamic relaxation numerical method. These results were found to be in good agreement with those available in the literature of this book and solved using the exact analytical solution. Therefore, a wide spectrum of comparisons between the dynamic relaxation numerical solutions and analytical exact solutions have been undertaken to demonstrate the accuracy of the DR program. The outcomes of these comparisons are found to be of acceptable accuracy. These results show that the convergence of the DR solution depends on several factors including the following:

1. Time increment:

It is a very important factor for speeding convergence and controlling numerical computations.

2. Damping coefficients:

Is that which produces critical motion.

3. Boundary conditions:

It is clear that the type of boundary condition is an important factor in determining the values of variables throughout the system.

4. Mesh size:

As the mesh size is reduced, the variables will be stable and smooth in values.

5. Discretization of elements:

Finer meshes reduce the discretization error, but at the same time increase the round off error due to the large number of calculations involved.

6. Fictitious densities:

They are used to evaluate the values at the far edges of the differential system. The fictitious densities vary from point to point over the system as well as for each iteration. Therefore, to stabilize the solution and to improve the convergence of the numerical computations fictitious densities must be applied.

With my best wishes

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